

Numerical simulation of the particle wall mass transfer rates on rough surfaces confining turbulent natural convection flows

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ABSTRACT

We performed numerical simulations of the particle mass transfer rates on the horizontal and vertical thermally active rough walls confining a turbulent natural convection flow. We considered the air flow conditions in a cubical cavity ($L = 1.22\text{ m}$, $Ra = 5.4 \cdot 10^8$, $Pr = 0.71$) with two opposed pairs of consecutive walls at different temperatures, for which measurements of the particle deposition velocities are reported in the literature. The simulations were carried out by solving the momentum, thermal energy and particle mass transfer equations using a series of two-dimensional computational cell units to determine spatial evolution and development of the particle concentration boundary layer in the vicinity of hot and cold rough walls. Different flow conditions, wall orientations, sizes of roughness elements, particle diameters and temperature increments between the wall and the bulk fluid, typically found in indoor environments, have been considered. At large mass transfer particle Péclet numbers, corresponding to spherical particles with diameters of $0.5\text{-}1\ \mu\text{m}$, the wall mass transfer rates of particles on rough textures typically increase by a factor of about 3 in comparison with the deposition on smooth surfaces. The thermophoretic effect significantly decreases the particle deposition on hot surfaces and increases it on cold surfaces, especially at large particle Péclet numbers. The numerical predictions of

31 the deposition velocities are in good agreement with the measurements reported in
32 the literature for the same flow conditions considered.

33

34 **NOMENCLATURE**

35	c	Concentration [kg m^{-3}]
36	C_D	Drag coefficient [-]
37	C_c	Cunningham slip correction factor [-]
38	d	Size of the roughness element [m]
39	d_p	Particle diameter [m]
40	D	Diffusion coefficient [$\text{m}^2 \text{s}^{-1}$]
41	g	Gravity acceleration [m s^{-2}]
42	h	Height of the computational domain [m]
43	k	Thermal conductivity [$\text{W m}^{-1} \text{K}^{-1}$]
44	k_B	Boltzmann constant [J K^{-1}]
45	k_c	Convection mass transfer coefficient [m s^{-1}]
46	Kn	Knudsen number ($Kn = 2 \lambda/d_p$) [-]
47	K_{TP}	Thermophoresis constant [-]
48	L	Length of the cubical cavity [m]
49	L_w	Length of the surface [m]
50	n	Coordinate perpendicular to the wall [m]
51	p	Pressure
52	Pe	Mass transfer Péclet number ($Pe = Re_d Sc$)
53	Pr	Prandtl number ($Pr = \nu/\alpha$) [-]
54	Ra	Rayleigh number ($Ra = g\beta\Delta TL^3/\nu\alpha$) [-]
55	Re	Reynolds number ($Re = U_\infty L/\nu$) [-]
56	Re_d	Reynolds number ($Re_d = S_u d^2/\nu$) [-]
57	Re_p	Particle Reynolds number ($Re = U_p - U_f d_p/\nu$) [-]
58	Sc	Schmidt number ($Sc = \nu/D$) [-]
59	Sh	Sherwood number ($Sh = k_c d/D$) [-]
60	S_T	Temperature gradient [K m^{-1}]
61	S_u	Shear rate [s^{-1}]
62	t	Time [s]

63	T	Temperature [K]
64	u, v	Velocity components [m s^{-1}]
65	U	Velocity magnitude [m s^{-1}]
66	x, y	Cartesian coordinates [m]
67	<i>Greek letters</i>	
68	α	Thermal diffusivity [$\text{m}^2 \cdot \text{s}^{-1}$]
69	β	Thermal expansion coefficient [K^{-1}]
70	δ	Boundary layer thickness [m]
71	Δ	Increment [-]
72	λ	Mean free path [m]
73	μ	Dynamic viscosity [$\text{kg} \cdot \text{m}^{-1} \text{s}^{-1}$]
74	ν	Kinematic viscosity [$\text{m}^2 \cdot \text{s}^{-1}$]
75	ρ	Density [$\text{kg} \cdot \text{m}^{-3}$]
76	τ	Time scale [s]
77	v	Deposition velocity [m s^{-1}]
78	<i>Subscripts</i>	
79	∞	Value outside the boundary layer
80	C	Cold
81	f	Fluid
82	g	Gravitational settling
83	h	Horizontal
84	H	Hot
85	in	Inlet
86	l	Linear
87	p	Particle
88	s	Settling
89	v	Vertical
90	w	Wall
91	<i>Subscripts</i>	
92	*	Non-dimensional
93	<i>Special symbols</i>	
94	$\langle \ \rangle$	Surface averaged

95 **1. INTRODUCTION**

96

97 Solid particle deposition onto surfaces holds significant relevance in numerous
98 industrial, technological, environmental, and medical applications. Examples of such
99 applications can be found in various fields such as the semiconductor industry (Simard-
100 Normandin, 2020), multiphase chemical reactors (Boccardo et al., 2019), indoor air
101 quality (Inthavong, 2020), soiling of artworks (Grau-Bové et al., 2019), and drug
102 delivery through dry powder inhalers (Shahin and Chablani, 2023).

103

104 In particular, the deposition of particles on both smooth and rough surfaces within
105 ducts and channels under forced convection air flows bears significant implications in
106 the aforementioned applications. Researchers including Zhao and Wu (2006), Dritselis
107 (2017), Wang et al. (2019), Hong et al. (2020) have numerically examined this flow
108 configuration, while Hahn et al. (1985), Kvasnak et al. (1993), and Chavez et al. (2020)
109 have conducted experimental studies. These investigations aimed to analyze the
110 impact of rough elements' geometry on particle interception and deposition.
111 Collectively, these studies consistently demonstrate an overall increase in particle
112 deposition rates on rough surfaces when compared to smooth surfaces.

113

114 The removal rate of particles by deposition is also important to predict the decay of
115 bulk concentration which, in turn, is key to assess the health risk due to particle
116 inhalation. (Abadie et al. 2001, Lai, 2002). The deposition rate of aerosol particles is
117 usually experimentally determined by monitoring the concentrations of airborne
118 aerosol particles or by measuring the amount of deposited particles on sample
119 surfaces installed on the walls adjacent to particle laden flows. Table I summarizes
120 some of the experimental studies analyzing particle deposition on rough surfaces using
121 test chambers. The existing studies in the field have predominantly focused on
122 conducting experiments within chambers under forced convection conditions created
123 by fans. In contrast, only Kalilainen et al. (2016) and Thatcher and Nazaroff (1997)
124 reported measurements of particle deposition under controlled natural convection
125 flows for smooth and rough surfaces respectively.

126

Reference	Chamber dimensions (m)*	Roughness height (microns)	Roughness shape	Particle diameters (microns)	Surface orientation	Heat transfer	Flow
Shimada et al. (1988)	Cylinder Diameter 0.15 Height 0.15	64-201	Stainless steel with structured roughness, Sandpaper	0.01-0.18	Sidewall	Isothermal	Fan
Thatcher and Nazaroff (1997)	1.22 x 1.22 x 1.22	10, 180, 2400	Scratched, dry wall texture, balls	0.1, 0.5, 1.3	Sidewalls, ceiling and floor	Hot sidewall and floor. Cold sidewall and ceiling	Natural convection
	4.9 x 3 x 2.4	10, 180, 2400	Scratched, dry wall texture, balls	0.2, 1	Hot and cold sidewalls	Hot and cold sidewalls	Natural convection
Abadie et al. (2001)	0.6 x 0.6 x 0.6	0.3-50.8	Wall papers, carpet, wood	0.7-5	Sidewalls, ceiling and floor	Isothermal	Fan
Lai et al. (2002)	2 x 2 x 2	Smooth, 2000	Smooth Perspex, Structured roughness	0.7-5.4	Sidewalls	Isothermal	Fan
Lai and Nazaroff (2005)	1.22 1x 1.22 x 1.22	10, 70, 100, 150, 250	Glass and sandpaper	0.9-9.1	Sidewalls	Isothermal	Fan
Hamdani et al. (2008)	0.5 x 0.5 x 0.5	1.5-26	Vinyl, wallpapers, wood, wall-plaster and concrete	0.35-2	Sidewalls	Isothermal	Fan
Hussein et al. (2009)	1 x 1 x 1	47-368	Wall papers, wall-plasters, carpets	0.02-5	Sidewalls, ceiling and floor	Isothermal	Fan
Kalilainen et al. (2016)	0.7 x 0.7 x 0.7	Smooth	-	1, 2.5	-	Hot and cold sidewalls	Natural convection
Wang et al. (2018)	0.8 x 0.8 x 0.8	64-198	textile materials	0.37-1.6	Sidewalls	Isothermal	Fan
Zhong et al. (2021)	0.3 x 0.3 x 0.3	500	3D printed rectangular, triangular, semi-circular ribs	0.3-5	Floor	Isothermal	Fan

127 (*) Length x width x height

128 Table I. Summary of experimental studies analyzing particle deposition on rough
129 surfaces installed on the walls of chambers.

130

131 Indoor natural convection flows produced by different temperatures of the confining
132 walls are typically encountered in poorly or unventilated spaces. Under these flow
133 conditions, with relatively low velocities in the bulk of the fluid and relatively large

134 temperature gradients near the thermally active walls, the deposition of particles
135 depends strongly on the orientation of the wall, favoring, because of gravity, the
136 landing of particles, with relatively large inertia, on horizontal surfaces facing up.
137 Another effect that can strongly influence the particle deposition rates on the
138 thermally active walls is the thermophoretic effect, which enhances the particle mass
139 flux toward cold walls and decreases it on hot walls (see for example Michaelides,
140 2006). This relatively complex scenario, involving gravitational, thermophoretic and
141 Brownian forces acting on the particles, can be further complicated by the effect of the
142 surface roughness elements, which are known to generally enhance the particle
143 deposition on the walls for forced convection conditions.

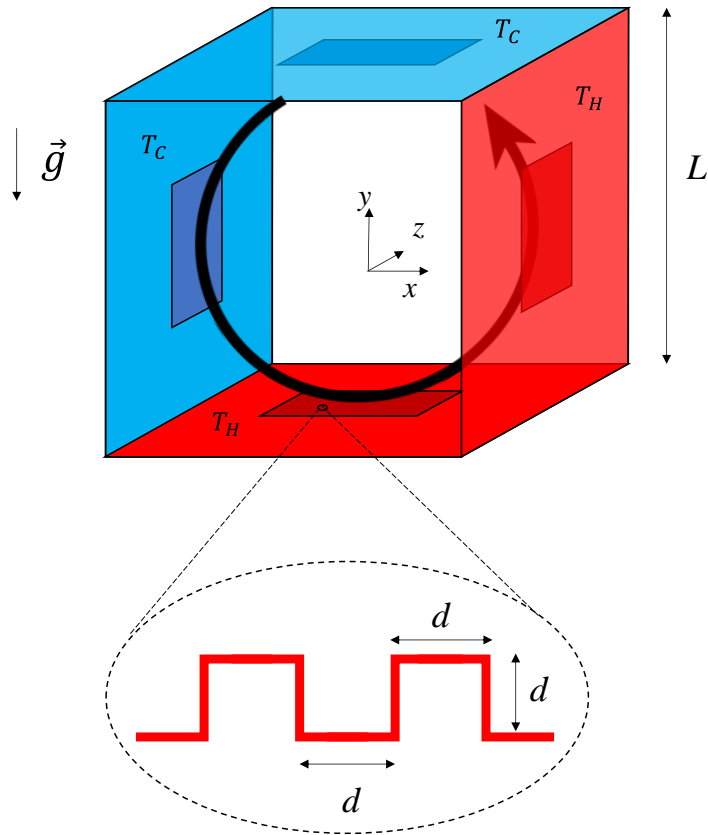
144

145 The objective of this study is to analyze the effect of the surface roughness elements
146 on the particle deposition rates on the thermally active walls of a confined turbulent
147 natural convection flow and contribute to improve our current limited understanding
148 of this flow configuration that has important implications for indoor air quality and
149 valuable surface soiling. We report numerical simulations for different particle
150 diameters, roughness sizes, surface orientations and wall-to-fluid temperature
151 increments, typically found in indoor environments and exhibition cases of artwork.

152

153

154 **2. PHYSICAL MODEL**



155

156

Figure 1. Physical model

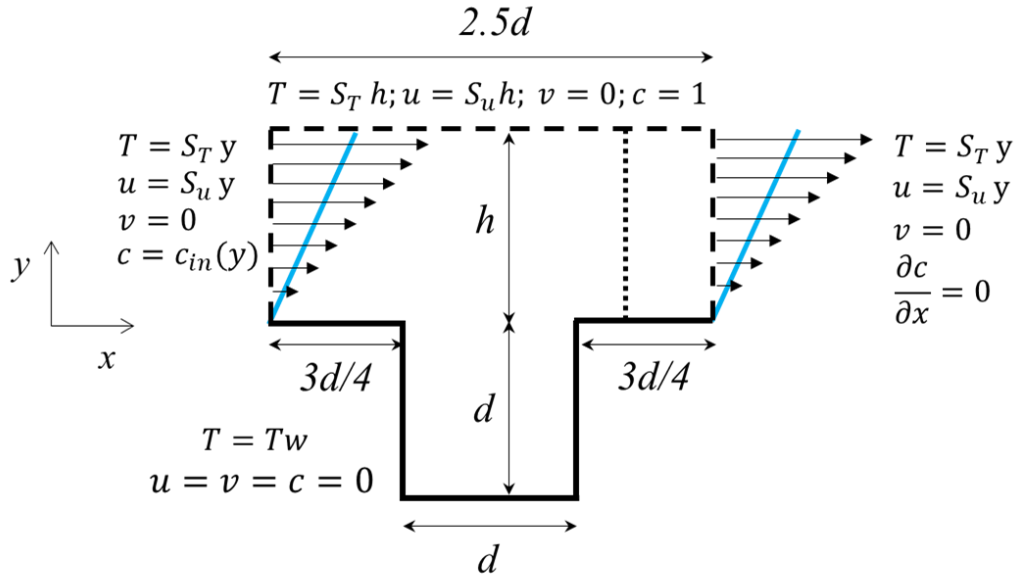
157 Figure 1 shows a sketch of the cubical cavity used in the experiments of Thatcher and
158 Nazaroff (1997), who measured particle deposition rates on rough surfaces using this
159 natural convection flow configuration. The flow is driven by heating, at constant
160 temperature T_H , the bottom wall and a vertical wall. The two opposite horizontal and
161 vertical walls are kept at a lower temperature, T_C . The remaining two vertical walls are
162 considered adiabatic. The flow at high Rayleigh numbers ($Ra \sim 10^8$) consists,
163 essentially, in a turbulent large-scale flow circulation and thin laminar boundary layers
164 attached to the thermally active walls (Fabregat and Pallares, 2020). In this study we
165 consider the particle deposition on rough surfaces installed on the central part of the
166 horizontal and vertical isothermal walls, mimicking the experimental arrangement by
167 Thatcher and Nazaroff (1997). The roughness elements are assumed to be two
168 dimensional with a square cross section (see Fig. 1).

169

170 As reported by Lai and Nazaroff (2005) the typical sizes of the small wall roughness
171 elements (d) found indoors range between $70 \mu\text{m}$ and $250 \mu\text{m}$. These sizes are three
172 to four orders of magnitude smaller than the typical sizes of display cases of artworks
173 or indoor environments ($L \sim 1\text{m}$). This relatively large ratio between length scales
174 ($L/d \sim 10^4 - 10^3$) allows us to assume that the development of the mass transfer
175 boundary layer over a distance equal to a few hundred roughness elements occurs
176 essentially under constant velocity and temperature profiles. To simulate the mass
177 transfer boundary layer over a surface covered by a large number of roughness
178 elements we considered a series of unit computational cells constituted by two
179 consecutive roughness elements. The computational strategy consists of sequentially
180 solving the concentration field in a unit computational cell using the outflow
181 concentration profile as the inflow boundary condition in the next downstream cell.
182 Compared to a single computational domain consisting of the totality of the roughness
183 elements, this strategy allows a significant reduction in computational costs. The
184 procedure of these sequential simulations is described in detail in the next section
185 where the computational aspects of the problem are discussed.

186

187 Direct Numerical Simulation (DNS) of the turbulent flow at $Ra = 5.4 \cdot 10^8$ and $Pr =$
188 0.71 in this flow configuration ($L = 1.22 \text{ m}$) revealed that the distance from the wall
189 for which the velocity and thermal boundary layers have a constant gradient (i.e. linear
190 profiles) is about $\delta_l \approx 0.01 L$ (see Figs. 13 and 14 of Fabregat and Pallares, 2020). This
191 simulation also indicates that the boundary layers near the vertical and horizontal
192 walls are laminar and essentially steady, with relatively small fluctuations intensities.
193 Consequently, the ratio between the thickness of these layers, characterized by linear
194 velocity and temperature profiles, and the typical size of the roughness elements
195 ($70 \mu\text{m} \leq d \leq 250 \mu\text{m}$) ranges between 170 and 50 for the smallest and largest
196 elements, respectively. Under these conditions, and taking into account the large
197 Schmidt number ($Sc \sim 10^4 - 10^6$) of the particles based on the Brownian diffusion
198 coefficient, it is reasonable to consider that the particle mass transfer boundary layer
199 develops within a region of constant velocity and temperature gradients (i.e. with
200 linear velocity and temperature profiles). We denote this constant shear rate as, S_u
201 and the constant temperature gradient as S_T .



202

203

Figure 2. Computational domain and boundary conditions.

204

205 According to these hypotheses, the two-dimensional physical domain considered for
 206 the simulations is shown in Fig. 2. Non-slip and constant and uniform temperature
 207 boundary conditions are imposed at the bottom boundary of the domain. In this wall,
 208 the concentration of the particles is set to zero according to the assumption of a
 209 perfect sink for particles (i.e. no rebound and no particle resuspension). At the inlet
 210 and outlet boundaries constant linear velocity and temperature profiles are
 211 considered. For the concentration we impose a known vertical concentration profile at
 212 the inlet and outflow conditions at the outlet. The mathematical details of these
 213 boundary conditions are given in next section. The top boundary, located at a distance
 214 (h) from the top of the roughness elements, is considered to be far enough from the
 215 edge mass transfer boundary to impose constant and uniform velocity, temperature
 216 and concentration.

217

218

219 3. MATHEMATICAL MODEL

220

221 3.1. Governing equations

222

223 Considering constant physical properties and steady flow conditions within the two-
224 dimensional boundary layer, the non-dimensional continuity, momentum and thermal
225 energy equations can be written as

226

$$227 \quad \frac{\partial u_i^*}{\partial x_i^*} = 0, \quad (1)$$

228

$$229 \quad \frac{\partial u_j^* u_i^*}{\partial x_j^*} = -\frac{\partial p^*}{\partial x_i^*} + \frac{1}{Re_d} \frac{\partial^2 u_i^*}{\partial x_j^{*2}} \quad (2)$$

230 and

$$231 \quad \frac{\partial u_j^* T^*}{\partial x_j^*} = \frac{1}{Re_d Pr} \frac{\partial^2 T^*}{\partial x_j^{*2}}. \quad (3)$$

232

233 The time and length scales, used to obtain the non-dimensional variables, are the
234 inverse of the shear rate S_u and the height of the roughness elements, d , consequently
235 the velocity scale is d/S_u . In Eq. 3, $Re_d = S_u d^2/\nu$ is the Reynolds number and $Pr =$
236 ν/α is the Prandtl number. The non-dimensional temperature is defined as $T^* =$
237 $(T - T_\infty)/|T_w - T_\infty|$. Note that according to this definition $T^* = 1$ at the hot wall
238 ($T_w > T_\infty$) and $T^* = -1$ at the cold wall ($T_w < T_\infty$).

239

240 The non-dimensional mass transfer equation that governs the steady state two-
241 dimensional spatial distribution of the concentration of particles can be expressed as,

242

$$243 \quad \frac{\partial u_{pi}^* c^*}{\partial x_i^*} = \frac{1}{Re_d Sc} \frac{\partial^2 c^*}{\partial x_i^{*2}}. \quad (4)$$

244

245 In Eq. 4, u_{pi}^* is the non-dimensional particle velocity and $Sc = \nu/D$ is the Schmidt
 246 number. The diffusion coefficient of the particles, D , is computed with the Stokes-
 247 Einstein equation for spherical particles

248

$$249 \quad D = k_B T_w C_c / 3\pi\mu d_p. \quad (5)$$

250

251 In Eq. 5, C_c is the Cunningham factor, which can be expressed as a function of the
 252 Knudsen number ($Kn = 2 \lambda/d_p$) as (Allen and Raabe, 1982)

253

$$254 \quad C_c = 1 + Kn[1.205 \exp(-0.0026/Kn) + 0.425 \exp(-0.74/Kn)]. \quad (6)$$

255

256 Considering the dilute approximation (i.e. no interaction between particles and
 257 particles do not influence the fluid velocity), the local particle velocity is governed by
 258 the second Newton's law, which for $Re_p < 1$, ($C_D = 24/Re_p$ i.e. small spherical
 259 particles in creeping flow) can be written as

260

$$261 \quad \frac{d\vec{v}_p}{dt} = \frac{(\vec{v} - \vec{v}_p)}{\tau_p} + \left(1 - \frac{\rho}{\rho_p}\right) \vec{g} - \frac{18\mu^2 C_c}{\rho \rho_p d_p^2} K_{tp} \frac{\vec{\nabla} T}{T_\infty}. \quad (7)$$

262

263 The different terms on the right-hand side of Equation 7 are, from left to right, the
 264 hydrodynamic drag force, the gravity force and the thermophoretic force. The
 265 accelerations originated from the pressure gradient force, the added mass force, the
 266 Basset force and the lift force, which are proportional to the ratio between the fluid
 267 density and the particle density are neglected in Eq. 7 since, for solid gas flows,
 268 $O(\rho/\rho_p) \sim 10^{-3}$. In Eq. 7, τ_p is the particle relaxation time defined in Eq. 8.

$$269 \quad \tau_p = \frac{\rho_p d_p^2}{18\mu C_c} \quad (8)$$

270

271 The value of K_{tp} in the thermophoretic force (see Eq. 7), for $Kn > 0.1$, can be
 272 computed as (Talbot et al. 1980)

273

$$274 \quad K_{tp} = \frac{2C_s(k+2k_p Kn)[1+2Kn(1.2+0.41 \exp(-0.44/Kn))]}{(1+6C_m Kn)(2k+k_p+4k_p C_t Kn)}, \quad (9)$$

275

276 with $C_s = 1.17$, $C_m = 1.14$ and $C_t = 2.18$.

277

278 Assuming small particles ($Re_p < 1$) and slow time varying fluid velocities near the wall,
279 within the laminar boundary layer, we can neglect the acceleration term on the left-
280 hand side of Eq. 7 and the non-dimensional velocity of the particle can be computed as

281

$$282 \quad u_{pi}^* = u_i^* + \left(1 - \frac{\rho}{\rho_p}\right) \frac{\tau_p g_i}{S_u d} - \frac{K_{tp}}{Re_d} \left| \frac{T_w}{T_\infty} - 1 \right| \frac{\partial T^*}{\partial x_i^*}. \quad (10)$$

283

284 In Equation 10, the different terms on the right-hand side are, from left to right, the

285 fluid velocity, the non-dimensional settling velocity, $V_{si}^* = \left(1 - \frac{\rho}{\rho_p}\right) \frac{\tau_p g_i}{S_u d}$, and the

286 thermophoretic velocity. The constant settling velocity can be absorbed into the fluid
287 velocity, ($\tilde{u}_i^* = u_i^* + V_{si}^*$, see for example Pallares and Fabregat, 2020).

288

$$289 \quad u_{pi}^* = \tilde{u}_i^* - \frac{K_{tp}}{Re_d} \left| \frac{T_w}{T_\infty} - 1 \right| \frac{\partial T^*}{\partial x_i^*}. \quad (11)$$

290

291 The introduction of the particle velocity (Eq. 11) into Eq. 4 results in

292

$$293 \quad \tilde{u}_i^* \frac{\partial c^*}{\partial x_i^*} = \frac{K_{tp}}{Re_d} \left| \frac{T_w}{T_\infty} - 1 \right| \frac{\partial}{\partial x_i^*} \left(c^* \frac{\partial T^*}{\partial x_i^*} \right) + \frac{1}{Re_d Sc} \frac{\partial^2 c^*}{\partial x_i^{*2}}. \quad (12)$$

294

295 Equation (12) is the non-dimensional mass transfer equation as a function of the local
296 fluid velocities and temperatures.

297

298 *3.2. Boundary conditions and computational model*

299

300 According to the hypotheses described above, Equations 1, 2, 3 and 12 constitute the
301 set of governing equations of the problem. The solution of these equations, together
302 with the corresponding boundary conditions, has been obtained sequentially in a
303 series of concatenated computational cells. The computational unit cell is shown in Fig.
304 2. The boundary conditions are indicated in this figure. For given values of the

305 Reynolds number and of the temperature gradient, the velocity and the temperature
306 distribution are the same in all the computational cells, according to the fully
307 developed assumption for velocity and temperature. At the first computational cell we
308 impose $c^* = 1$ at the inlet of the domain and zero diffusion flux at the outlet (see Fig.
309 2). Once the concentration field is converged in the first computational cell, the
310 concentration profile at $x^* = 2$, indicated in Fig. 2 with a dotted vertical line, is used as
311 inlet boundary conditions for the second computational cell. This procedure is
312 repeated sequentially for 1000 computational cells. The streamwise extension of the
313 computational cell is $L_x^* = 2.5$ (see Fig. 2) and the concentration profile to be imposed
314 to the next computational cell is taken at $x^* = 2$ to minimize the effect of the outlet
315 boundary condition for concentration on the concentration distribution at the inlet of
316 the next computational cell.

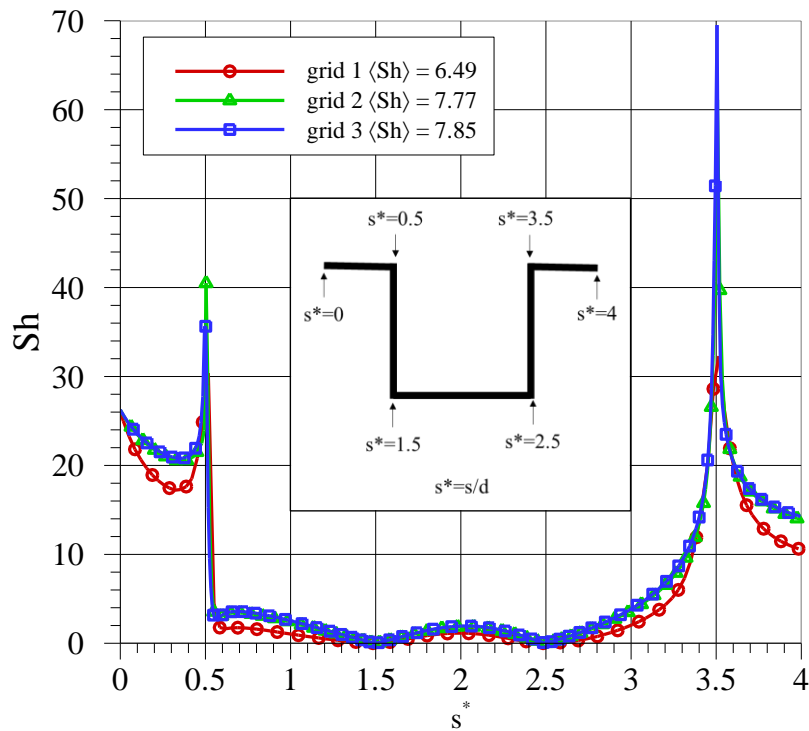
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318 Equations 1, 2, 3 and 12 and the corresponding boundary conditions were solved
319 numerically with the second-order accuracy finite volume code 3DINAMICS. For the
320 momentum and thermal energy equations the discretization of the convective terms is
321 centered and for the mass transfer equation is based on the TVD scheme (Koren,
322 1993), which avoids the non-physical wiggles that appear at large Péclet numbers
323 produced by the centered scheme. The diffusion terms of all equations are discretized
324 with the centered second order scheme. This code has been used successfully for the
325 simulation of mass transfer boundary layer flows (Pallares and Grau 2014, Pallares and
326 Ferré 2015, Pallares 2015) and of convection heat transfer in porous media using the
327 unit cell scale approach (Pallares and Grau, 2010).

328

329 To select the grid resolution, we performed a simulation in the first computational cell,
330 which is the one with the largest concentration gradients, using three different meshes
331 at the largest Péclet number considered ($Pe = Re_d Sc \approx 3 \cdot 10^4$). All the grids have the
332 nodes stretched towards the walls. The computational cell has dimensions 2.5×17 ($h =$
333 $16 d$, see Fig. 2). This vertical dimension is larger than the mass transfer boundary
334 layer thickness at the last computational cell for the lowest Péclet number considered
335 ($Pe \approx 26$). The coarse grid (grid 1) has 73×218 grid points, the medium grid, 143×368
336 grid points and the fine grid, 187×434 grid points. Figure 3 shows the numerical

337 predictions of the local Sherwood number ($Sh = \frac{\partial c^*}{\partial n^*} \Big|_w$) along the wall for the three
 338 meshes. It can be seen that grid 2 and grid 3 give very similar results. The difference
 339 between the surface averaged Sherwood numbers of these two grids is about 1% for
 340 these extreme conditions (i.e. largest Péclet number and first computational cell)
 341 corresponding to the largest Sherwood numbers. According to this, grid 2 has been
 342 selected for the simulations.



343
 344 Figure 3. Grid independence test. Local Sherwood number along the wall. Surface
 345 averaged values are indicated in the legend.

347 3.3. Parameters of the simulations

348
 349 Table II indicates the parameters of the experimental flow configuration used by
 350 Thatcher et al. (1996) and Thatcher and Nazaroff (1997) to carry out the
 351 measurements of particle deposition on smooth and rough horizontal and vertical
 352 surfaces produced by a confined turbulent natural convection of air using the flow
 353 configuration shown in Fig. 1. We included in Table II, the Reynolds number, the non-
 354 dimensional wall shear rates [$S_u^* = (L/U_\infty)(\partial u/\partial y)$] and the temperature gradients

355 $[S_T^* = (2L/\Delta T)(\partial T/\partial y)]$ at the center ($x/L = 0.5$) of the horizontal (S_{uh}^* and S_{Th}^*) and
356 vertical (S_{uv}^* , S_{Tv}^*) walls computed from the DNS of Fabregat and Pallares (2020). Note
357 that the temperature scale is half of the temperature increment between the hot and
358 cold walls (i.e. $\Delta T = T_H - T_C$; $|T_w - T_\infty| = \Delta T/2$).

359

L (m)	ΔT (K)	Ra	Pr	Re $U_\infty L/\nu$	S_{uh}^*, S_{uv}^* $\frac{x}{L} = 0.5$	S_{Th}^*, S_{Tv}^* $\frac{x}{L} = 0.5$
1.22	3	$5.4 \cdot 10^8$	0.70	6640	53.2, 143.9	-38.3, -45.9

360 Table II. Non-dimensional numbers corresponding to the experimental conditions of
361 Thatcher and Nazaroff (1997).

362

363 Table III summarizes the ranges of the different parameters of the particles considered
364 for the simulations. The physical properties and sizes of the particles correspond to the
365 ammonium fluorescein particles used by Thatcher and Nazaroff (1997) in their
366 measurements of particle deposition on rough surfaces. These rough surfaces with
367 length $L_w = 0.1$ m and width 0.3 m were installed on the central part of the horizontal
368 and vertical thermally active walls of a test chamber as depicted in Fig. 1.

369

d_p (μm)	ρ_p (kg/m^3)	k_p ($\text{W}/\text{m}\cdot\text{K}$)	K_{tp}	Kn	C_c	D (m^2/s)	Sc
0.1	1350	0.430	0.247	1.36	2.97	$6.69 \cdot 10^{-10}$	$2.24 \cdot 10^4$
0.5			0.240	0.272	1.33	$6.69 \cdot 10^{-11}$	$2.50 \cdot 10^5$
1.3			0.227	0.105	1.12	$1.95 \cdot 10^{-11}$	$7.71 \cdot 10^5$

370 Table III. Physical parameters of the particles considered for the simulations.

371

372 We selected the minimum and maximum sizes of the typical roughness elements
373 found indoors according to Lai and Nazaroff (2005). ($d=70$ μm and $d=250$ μm) and the
374 size of the roughness elements ($d=180$ μm) used by Thatcher and Nazaroff (1997).

375 Table IV indicates the values of the Reynolds numbers based on the size of the

376 roughness elements and on the wall shear rate on the horizontal (Re_{dh}) and vertical
 377 (Re_{dv}) walls. The corresponding non-dimensional wall shear rates are also included.

378

d (μm)	L/d	Re_{dh} $S_{uh}d/\nu$	Re_{dv} $S_{uv}d/\nu$	S_{Th}^{**} S_{Th}^*d/L	S_{Tv}^{**} S_{Tv}^*d/L
70	$1.74 \cdot 10^4$	$1.16 \cdot 10^{-3}$	$3.15 \cdot 10^{-3}$	$-2.20 \cdot 10^{-3}$	$-2.63 \cdot 10^{-3}$
180	$6.78 \cdot 10^3$	$7.69 \cdot 10^{-3}$	$2.08 \cdot 10^{-2}$	$-5.65 \cdot 10^{-3}$	$-6.77 \cdot 10^{-3}$
250	$4.88 \cdot 10^3$	$1.48 \cdot 10^{-2}$	$4.01 \cdot 10^{-2}$	$-7.84 \cdot 10^{-3}$	$-9.41 \cdot 10^{-3}$

379 Table IV. Non-dimensional numbers corresponding to different roughness sizes.

380

381 It can be seen that the typical Reynolds numbers range from, approximately, 10^{-3} to
 382 $5 \cdot 10^{-2}$ corresponding, respectively, to the smallest and largest roughness elements.

383

384 Table V summarizes the non-dimensional parameters of the governing equations (Eqs.
 385 1-3 and 12) for the 60 different cases simulated. The simulations were carried out for
 386 four different Reynolds numbers, in the range of the values shown in Table IV, three
 387 different Schmidt numbers, corresponding to the three different particle sizes
 388 indicated in Table IV, and five different wall-to-fluid temperature increments. Note
 389 that each case has been computed in 1000 unit cells and this corresponds to 60000
 390 simulations in the computational domain of the unit cell shown in Fig. 2.

391

Re_d	Sc	Pe	$\frac{K_{tp}}{Re_d} \left \frac{\Delta T_{1/2}}{T_\infty} \right $		
			$\Delta T_{1/2} = 0$	$\Delta T_{1/2} = \pm 1.5^\circ\text{C}$	$\Delta T_{1/2} = \pm 3^\circ\text{C}$
$1.16 \cdot 10^{-3}$	$2.24 \cdot 10^4$	$2.60 \cdot 10^1$	0	$1.06 \cdot 10^0$	$2.12 \cdot 10^0$
	$2.50 \cdot 10^5$	$2.90 \cdot 10^2$		$1.03 \cdot 10^0$	$2.07 \cdot 10^0$
	$7.71 \cdot 10^5$	$8.94 \cdot 10^2$		$9.74 \cdot 10^{-1}$	$1.95 \cdot 10^0$
$7.69 \cdot 10^{-3}$	$2.24 \cdot 10^4$	$1.72 \cdot 10^2$		$2.23 \cdot 10^{-1}$	$4.46 \cdot 10^{-1}$
	$2.50 \cdot 10^5$	$1.92 \cdot 10^3$		$2.17 \cdot 10^{-1}$	$4.34 \cdot 10^{-1}$
	$7.71 \cdot 10^5$	$5.93 \cdot 10^3$		$2.05 \cdot 10^{-1}$	$4.10 \cdot 10^{-1}$
$2.08 \cdot 10^{-2}$	$2.24 \cdot 10^4$	$4.66 \cdot 10^2$		$5.94 \cdot 10^{-2}$	$1.19 \cdot 10^{-1}$
	$2.50 \cdot 10^5$	$5.20 \cdot 10^3$		$5.78 \cdot 10^{-2}$	$1.16 \cdot 10^{-1}$
	$7.71 \cdot 10^5$	$1.60 \cdot 10^4$		$5.45 \cdot 10^{-2}$	$1.09 \cdot 10^{-1}$
$4.01 \cdot 10^{-2}$	$2.24 \cdot 10^4$	$8.98 \cdot 10^2$		$3.07 \cdot 10^{-2}$	$6.15 \cdot 10^{-2}$
	$2.50 \cdot 10^5$	$1.00 \cdot 10^4$		$2.99 \cdot 10^{-2}$	$5.99 \cdot 10^{-2}$
	$7.71 \cdot 10^5$	$3.09 \cdot 10^4$		$2.82 \cdot 10^{-2}$	$5.65 \cdot 10^{-2}$

393

Table V. Simulated cases.

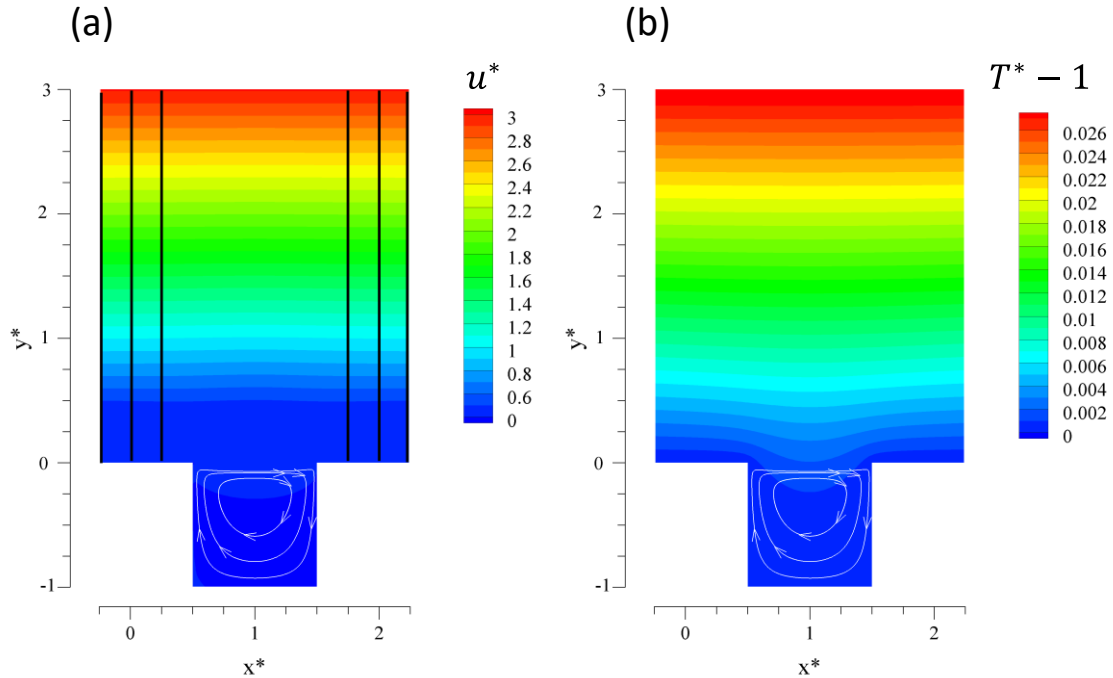
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395

396 4. RESULTS AND DISCUSSION

397

398 4.1. Velocity and temperature distributions



399

400 Figure 4. Velocity (a) and temperature (b) contours at $Re = 4.01 \cdot 10^{-2}$

401

402 As an example of the velocity and temperature distributions in a computational cell,

403 Figure 4 shows contours of the parallel-to-the-wall velocity component (Fig. 4.a) and

404 temperature (Fig. 4.b) at the largest Reynolds number considered ($Re = 4.01 \cdot 10^{-2}$)

405 near a cold wall. Note that the extension of an individual computational cell is

406 $L_x^* \times L_y^* = 2.5 \times 17$ and consequently Figure 4 shows only the contours near the wall

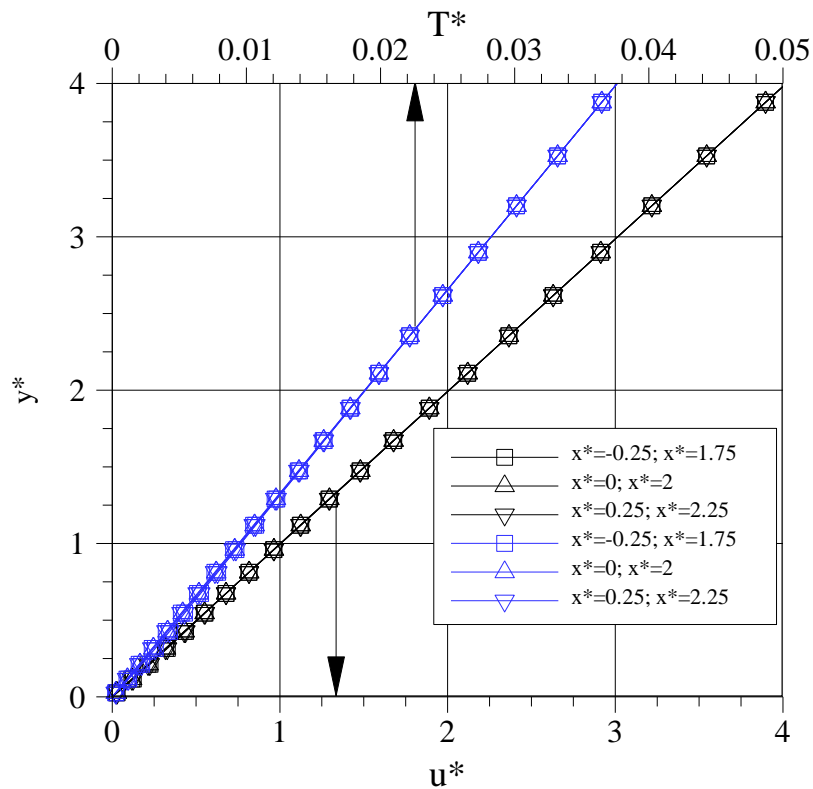
407 (i.e. for $y^* < 3$). The flow recirculation, depicted with streamlines in Figure 4, between

408 two consecutive roughness elements is highly symmetric according to the creeping

409 flow conditions ($Re \ll 1$). The temperature and velocity profiles along the vertical lines

410 indicated in Figure 4.a are plotted in Figure 5. The lines are located at $x^* =$

411 $-0.25, 0, 0.25, 1.75, 2$ and 2.25 .



412

413 Figure 5. Non-dimensional temperature and x-velocity component along the lines

414

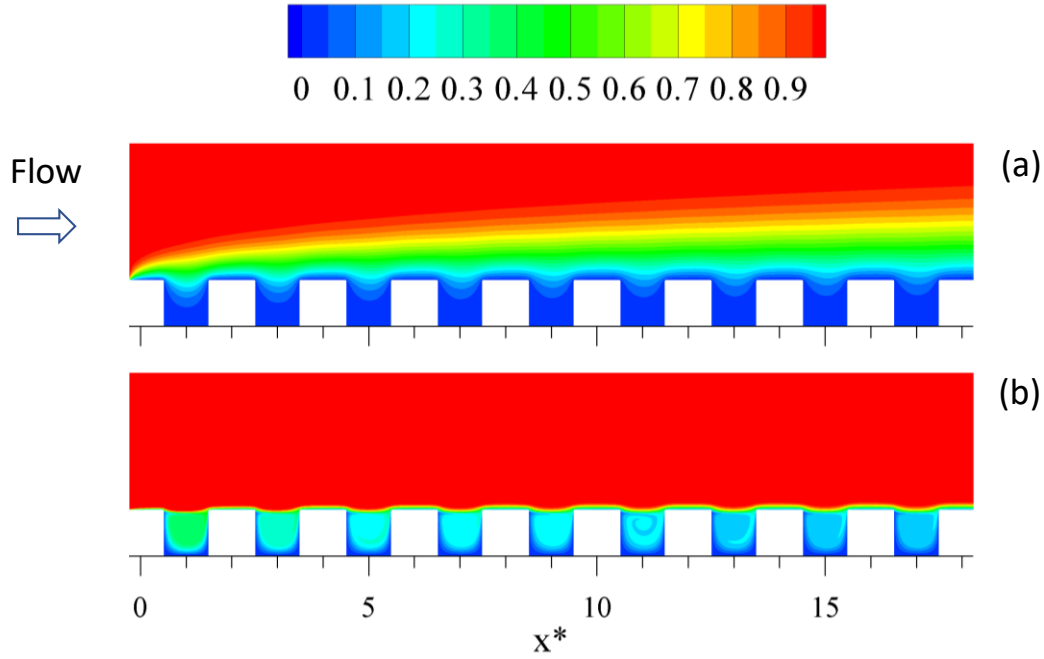
indicated in Fig. 4.a ($Re = 4.01 \cdot 10^{-2}$).

415

416 It can be seen that above the roughness elements the velocity and temperature
 417 distributions exhibit a vertical linear profile irrespectively of the streamwise position.

418 Also, it is evident that at these low Reynolds numbers the distortion of the velocity and
 419 temperature is located, near the wall, above the region between the roughness
 420 elements.

421 Figure 6 shows, for reference, the non-dimensional particle concentration contours,
 422 near the wall ($-1 < y^* < 3$) in the 10 first computational cells at $Pe = 2.60 \cdot 10^2$ (Fig.
 423 6.a) and $Pe = 5.20 \cdot 10^3$ (Fig. 6.b). The effect of the Péclet number is evident when
 424 comparing the mass transfer boundary layer thickness of the two cases depicted in this
 425 figure.



426

427 Figure 6. Non-dimensional particle concentration contours c and x -velocity at (a) $Pe =$
 428 $2.60 \cdot 10^2$ and (b) $Pe = 5.20 \cdot 10^3$.

429

430 4.2. Particle wall mass transfer rates and deposition velocities

431

432 In this section we report the particle wall mass transfer rate for the cases considered in
 433 Table V. The definition of the local Sherwood numbers based on the local mass
 434 transfer coefficient and the length of the wall (L_w) or the dimension of the roughness
 435 elements (d) are

436

$$437 \quad Sh_{L_w}(x) = \frac{k_c(x)L_w}{D}, \quad Sh_d(x) = \frac{k_c(x)d}{D}, \quad Sh_{L_w}(x) = Sh_d(x) \frac{L_w}{d} \quad (13)$$

438

439 The local Sherwood number, or the non-dimensional transfer rate, of a passive scalar
 440 from a smooth wall under a momentum boundary layer with a uniform and constant
 441 linear velocity profile can be predicted with the L  v  que solution (L  v  que, 1928).

442

$$443 \quad Sh_{L  v  que}(x) = \frac{k_c(x)L_w}{D} = 0.5384 \left(\frac{Re_d Sc}{x/d} \right)^{1/3} \frac{L_w}{d} \quad (14)$$

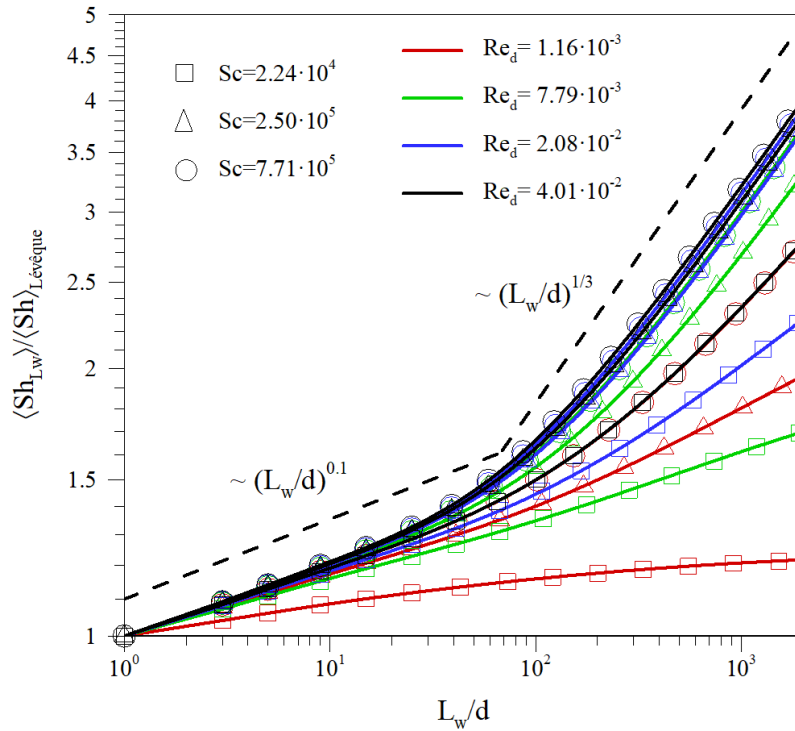
444

445 The corresponding surface averaged Sherwood can be written as

446

447
$$\langle Sh \rangle_{L\acute{e}v\hat{e}que} = \frac{\langle k_c \rangle L_w}{D} = 0.8075 (Re_d Sc)^{1/3} \left(\frac{L_w}{d} \right)^{2/3}$$
 (15)

448



449

450 Figure 7. Surface averaged Sherwood number, scaled with the Sherwood number
 451 corresponding to the L\acute{e}v\hat{e}que solution (smooth wall), as a function of the length of the
 452 surface with the roughness elements.

453

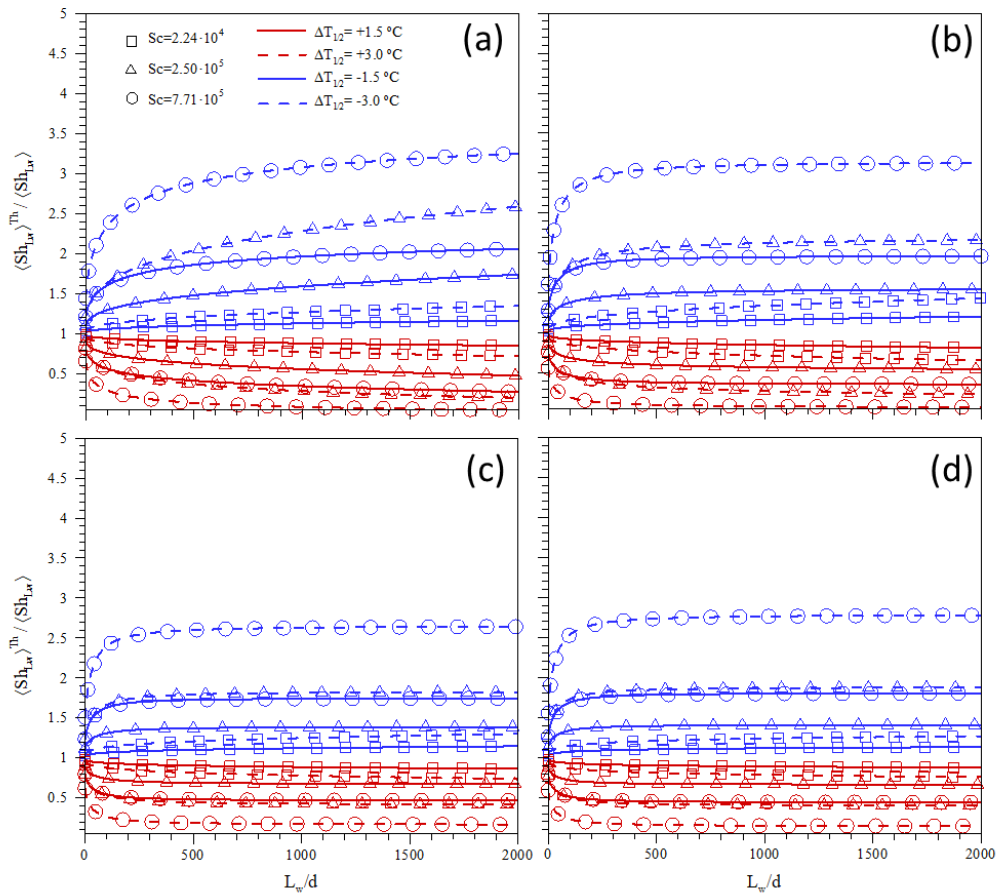
454 Figure 7 shows the ratio of the numerically predicted surface averaged Sherwood
 455 number and the surface averaged Sherwood number of a smooth surface according to
 456 the analytical solution proposed by L\acute{e}v\hat{e}que (1928).

457
$$\frac{\langle Sh_{L_w} \rangle}{\langle Sh \rangle_{L\acute{e}v\hat{e}que}} = \frac{\langle Sh_d \rangle}{0.8075 (Re_d Sc)^{1/3} \left(\frac{L_w}{d} \right)^{2/3}}$$
 (16)

458

459 It can be seen that the roughness elements increase the wall mass transfer rates with
 460 respect to the smooth wall. The augmentation depends on the product of the Reynolds
 461 number and the Schmidt number or the Péclet number. Note for example that the
 462 cases $Re_d = 4.01 \cdot 10^{-2}$, $Sc = 2.24 \cdot 10^4$ (black squares) and $Re_d = 1.16 \cdot 10^{-3}$, $Sc =$
 463 $7.71 \cdot 10^5$ (red circles) show almost coincident curves because of the very similar
 464 Péclet numbers ($Re_d Sc = 898.2$ and 894.4 , respectively). At high Péclet number
 465 ($Re_d Sc \rightarrow \infty$) the ratio of the Sherwood numbers increases with the one-third power
 466 of the length of the rough wall, for walls with lengths corresponding to more than
 467 approximately one hundred of roughness elements.

468



469

470

471 Figure 8. Surface averaged Sherwood number considering the thermophoretic effect,

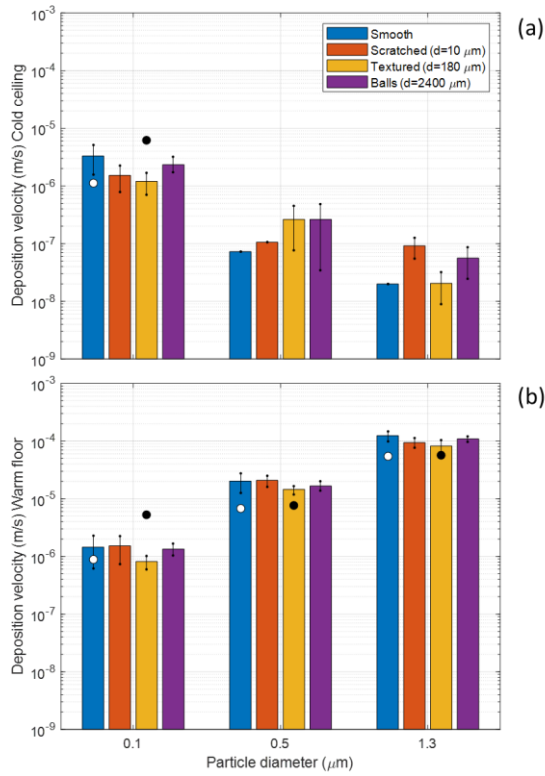
472 scaled with the Sherwood number corresponding to the isothermal case, as a function

473 of the length of the surface with the roughness elements. (a) $Re_d = 1.16 \cdot 10^{-3}$, (b)

474 $Re_d = 7.79 \cdot 10^{-3}$, (c) $Re_d = 2.08 \cdot 10^{-2}$, (d) $Re_d = 4.01 \cdot 10^{-2}$

475

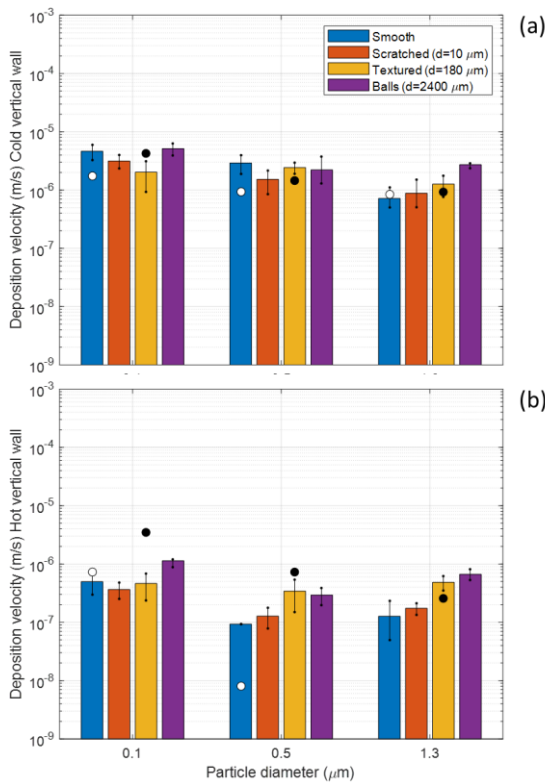
476 The effect of thermophoresis on the surface averaged Sherwood number is shown in
477 Figure 8. In this figure we plotted the ratio between the non-dimensional mass transfer
478 rate considering thermophoresis and the Sherwood number of the isothermal case
479 $(\langle Sh_L \rangle^{Th} / \langle Sh_L \rangle)$ as a function of the length of the roughness wall. As expected, Figure
480 8 shows that the deposition is enhanced, with respect to the isothermal case, on rough
481 cold walls (blue curves) and it is reduced on the hot walls (red curves). At high
482 Reynolds numbers (Figs. 8c to 8d) the ratio of the surface averaged Sherwood numbers
483 attains a constant value independently of the dimension of the rough wall for wall
484 lengths larger than approximately a few hundreds of rough elements ($L_w/d \approx 500$).
485 This ratio approximately corresponds to the size of one of the roughness elements
486 ($d = 180 \mu m$) and the length of the surface ($L_w = 0.1 m$) used in the experiments of
487 Thatcher and Nazaroff (1997) ($L_w/d = 555.6$). The increase of the deposition rate on
488 cold walls is more evident at low Reynolds number and high Schmidt number. For
489 example, at $Sc = 7.71 \cdot 10^5$, for a temperature increment of $\Delta T_{1/2} = -3.0^\circ C$ and the
490 lowest Reynolds number (Fig. 8a) the ratio of the Sherwood numbers is about 3 while
491 for the highest Reynolds number (Fig. 8d) is reduced down to 2.6. At low Reynolds
492 number and for the maximum temperature increment ($\Delta T_{1/2} = 3.0^\circ C$), the deposition
493 is essentially suppressed on the hot walls, especially for high Schmidt numbers.



494

495

Figure 9. Deposition velocities on the (a) cold ceiling and (b) warm floor.



496

497

Figure 10. Deposition velocities on the (a) cold and (b) hot sidewalls.

498

499

500 Finally, we compare the predictions of the present numerical simulations with the
501 experiments of Thatcher and Nazaroff (1997). These authors reported deposition
502 velocities on smooth and rough surfaces installed on the thermally active walls of the
503 cubical cavity sketched in Figure 1. One of the rough surfaces used in the experiments
504 was prepared by lightly scratching smooth cold-rolled stainless steel with a wire
505 grinding brush. The corresponding estimated size of the roughness elements for this
506 scratched surface is about $10\ \mu\text{m}$. A second surface was a painted aluminum casting of
507 skip-coat texture drywall taken from a residence resulting in a measured roughness
508 height of $180\ \mu\text{m}$. A third surface was a smooth surface with a mounted array of balls
509 with a diameter of $2400\ \mu\text{m}$. The corresponding Reynolds numbers (Re_d), based on the
510 shear rate on the horizontal walls, for these three surfaces are $2\cdot 10^{-5}$, $8\cdot 10^{-3}$ and 1.4,
511 respectively. It can be seen that only the second surface is within the range of
512 Reynolds numbers analyzed (see Table V). Figures 9 and 10 show the measured
513 deposition velocities for three different particle sizes (0.1 , 0.5 and $1.3\ \mu\text{m}$) with bars
514 and the model predictions with circles. Figure 9 corresponds to the horizontal walls
515 (cold ceiling and warm floor) and Figure 10, to the hot and cold sidewalls. The
516 deposition velocities on the horizontal walls have been computed by adding or
517 subtracting the terminal velocity of the particles ($v = k_c \pm V_s$) in the case of the
518 deposition on the floor or on the ceiling, respectively (Pallares and Fabregat, 2020). In
519 these figures we included the measurements of the particle deposition, reported by
520 Thatcher and Nazaroff (1997), on the scratched surface ($d = 10\ \mu\text{m}$) and on the
521 surface with balls ($d = 2400\ \mu\text{m}$) for comparison purposes. The predictions for
522 smooth surfaces obtained with the model, based on the boundary layer theory,
523 reported by Pallares and Fabregat (2021), are plotted with white circles and the
524 estimates corresponding to the present numerical simulations for $d = 180\ \mu\text{m}$ are
525 indicated with black circles.

526 Figure 9, corresponding to the horizontal walls, shows a larger deposition on the hot
527 floor (Fig. 9b) than in the cold ceiling (Fig. 9a) for the particles with diameters $0.5\ \mu\text{m}$

528 and 1.3 μm . The terminal velocities for these particles are $7 \cdot 10^{-6}$ m/s and $6 \cdot 10^{-5}$ m/s.
529 These values are close to the theoretically predicted and measured deposition
530 velocities on the hot wall indicating that the dominant deposition mechanism is
531 gravity. It can be seen that for these particles the measured and the predicted
532 deposition velocities are essentially independent of the rugosity of the surface. For the
533 smallest particles the terminal velocity is 10^{-7} m/s and the measured and theoretically
534 predicted deposition velocities are about one order of magnitude larger, $O(10^{-6})$ with
535 larger deposition on the cold ceiling than on the hot floor because of the
536 thermophoretic effect.

537

538 The model predicts no deposition on the cold ceiling (Fig. 9a) for the larger particles
539 independently of the rugosity of the wall because the terminal velocity of these
540 particles is larger than the predicted deposition velocities. As for the hot floor, the
541 theoretical prediction of the deposition velocity for the smallest particles is larger for
542 the rough wall than for the smooth surface. However, the measurements do not show
543 a robust trend considering the extension of the error bars.

544

545 Figure 10, corresponding to the vertical sidewalls, shows that the deposition velocities
546 are in general larger on the cold (Fig. 10a) wall than on the hot (Fig. 10b) wall, because
547 of the thermophoretic effect. On the cold sidewall the theoretical predictions are in
548 agreement with measurements and indicate that the deposition velocity is larger in the
549 rough surface than in the smooth wall. The boundary layer model predicts a very small
550 deposition or no deposition at all on the smooth cold sidewalls for the particles with
551 diameters of 0.5 μm and 1.3 μm (see Fig. 10b). It can be seen that there is a significant
552 increase of the deposition velocities for the textured rough surface in comparison with
553 the smooth surface, especially for the hot vertical wall.

554

555 **5. CONCLUSIONS**

556

557 We reported simulations of the particle wall mass transfer rates on heated or cooled
558 rough surfaces. The momentum and the thermal energy equations have been solved

559 numerically in the vicinity of the wall together with a mass transfer transport equation
560 for the particle concentration. To predict the particle deposition on relatively long
561 rough surfaces we have computed sequentially the mass transfer boundary layer
562 development in a series of unit computational cells with two consecutive two-
563 dimensional square roughness elements. The concentration of particles at the outlet of
564 an upwind unit computational domain is used as the boundary condition at the inlet of
565 the next unit computational cell. Following this strategy, we simulated the particle
566 deposition on rough surfaces with a length corresponding to 2000 roughness heights
567 periodically distributed along the streamwise direction for different flow conditions
568 (Reynolds numbers), temperature increments (thermophoretic effects) and particle
569 diameters (Schmidt number). The particle wall mass transfer rates on rough surfaces
570 increase in comparison with the smooth surfaces specially for large Péclet numbers
571 with typical factors between 3 and 4, for the conditions investigated. When the
572 thermophoretic effect is considered, the particle deposition is enhanced on the cold
573 walls with maximum factors of about 3 for large Péclet numbers while is essentially
574 suppressed on the hot walls. The numerical predictions of the particle deposition
575 velocities have been compared with measurements reported in the literature in a
576 cubical cavity with two pairs of opposed consecutive walls at different temperatures.
577 In general, both predictions and measurements, indicate that the deposition velocities
578 increase for rough surfaces when compared to smooth surfaces, especially for vertical
579 hot walls. On the horizontal walls the effect of the roughness is reduced, with
580 deposition velocities essentially independent of the rugosity of the wall, especially for
581 the large particles.

582

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584

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